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Und zurück.**

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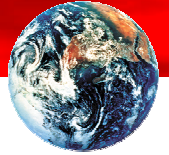
Space Qualification of a Fiber Optic Sensor System for X-38 Vehicle

Arnd Reutlinger

Project Manager Fiber Optic Systems

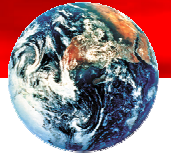
Oktober 5th, 2005



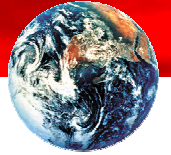


Content

- Introduction into Fiber Optic Sensing
- System Architecture
- Application on X-38
- Critical Technology Developments
- Other Applications and FOS Systems
- Summary

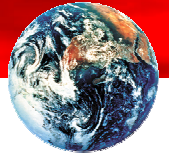


Introduction into Fiber Optic Sensing

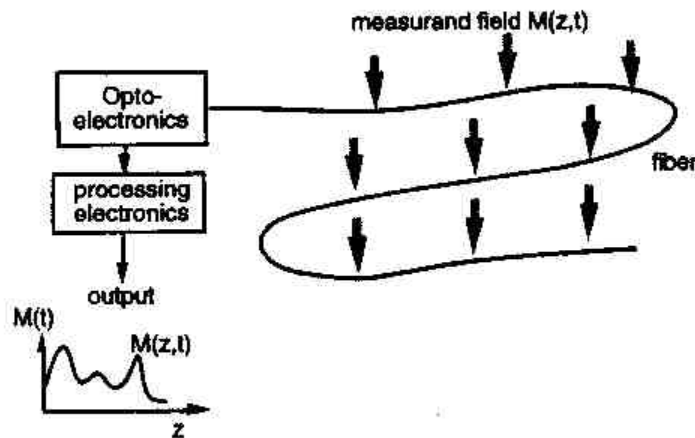


Fiber Optic Sensing – The advantages

- Simultaneous measurement of various parameters:
 - Temperature from -260 °C to 800 °C
 - Strain up to 50.000 $\mu\text{m}/\text{m}$ and load cycles $> 10^8$
 - Concentration in gases and fluids (i.e. hydrogen)
- Insensitive to extreme environments (EMI, loads, chemical compatibility)
- Small dimensions and low system weight, i.e. signal processing unit, 100 measurands and harness $< 5\text{kg}$
- Feasibility of structural conform embedding into composites
- Signal processing unit and sensor fiber qualified for deployment in extreme environments



Functional Principle



Bragg Grating

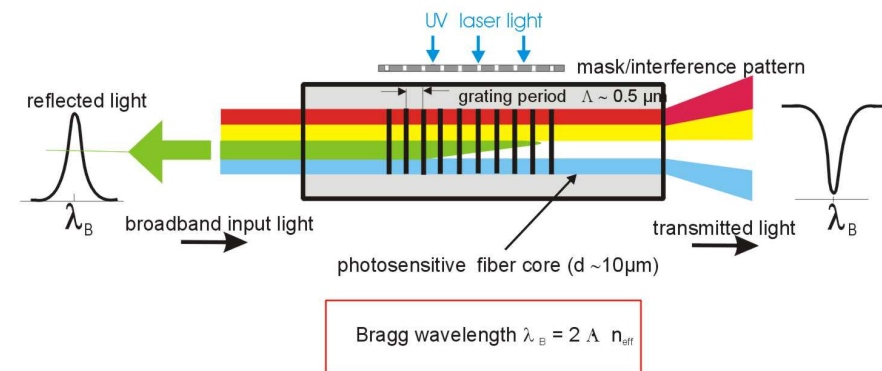
- Filtering of a small portion of light out of a broadband spectrum
- Inscription of Bragg gratings into the fiber core via a UV laser

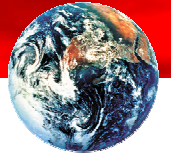
- Intrinsic distributed measurement points in an optical fiber
- Up to 20 Bragg gratings distributed along one single fiber
- Identification of single measurement points via WDM (wavelength division multiplexing)

Fiber Bragg Grating (FBG)

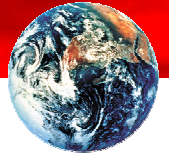
Micro optic key component for photonic applications in

- **Communication systems** : spectral filters and multiplexers (WDM),...
- **Sensor networks** : strain & vibration and temperature monitoring
→ optical counterpart to electrical strain gauges
- **Laser resonators** : narrowband reflectors for fiber & diode lasers

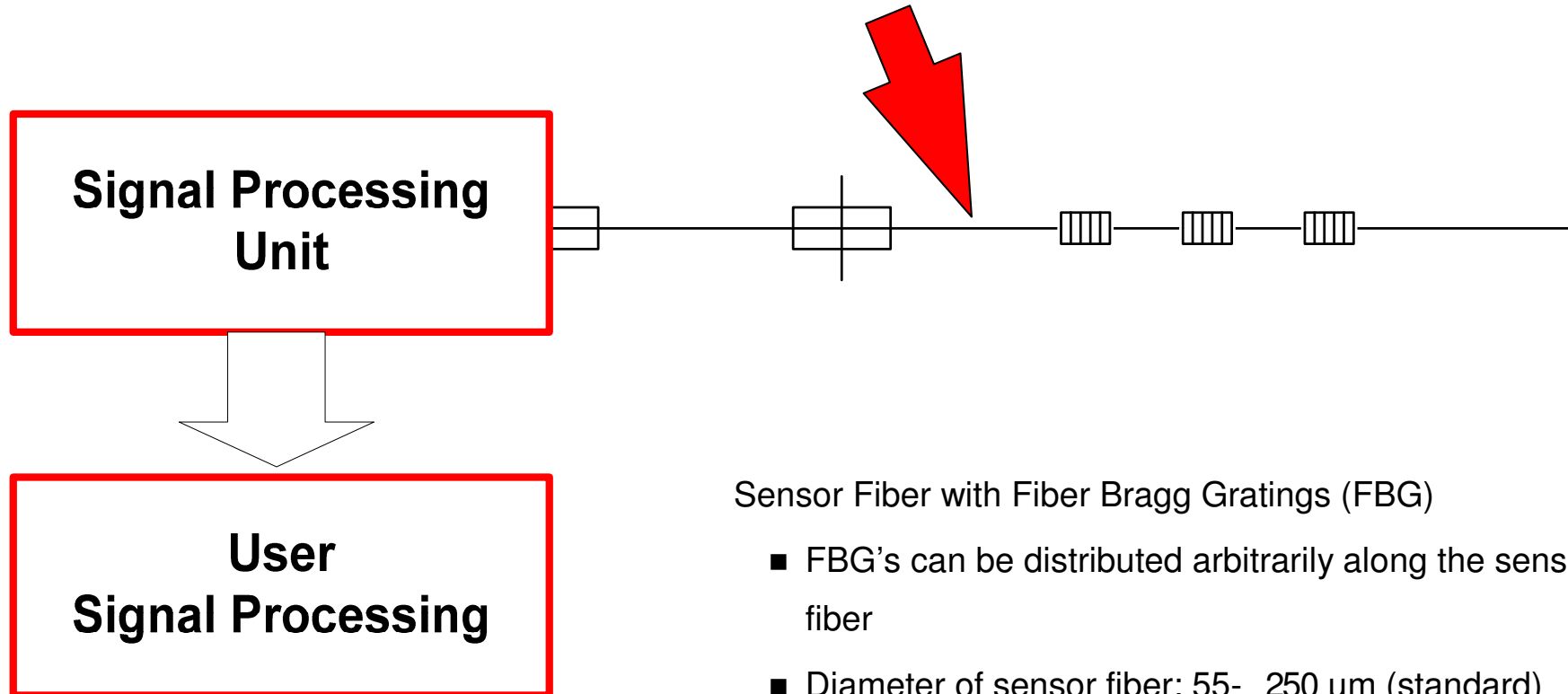




System Architecture

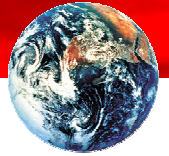


System Architecture – Sensor Faser and Fiber Bragg Gratings

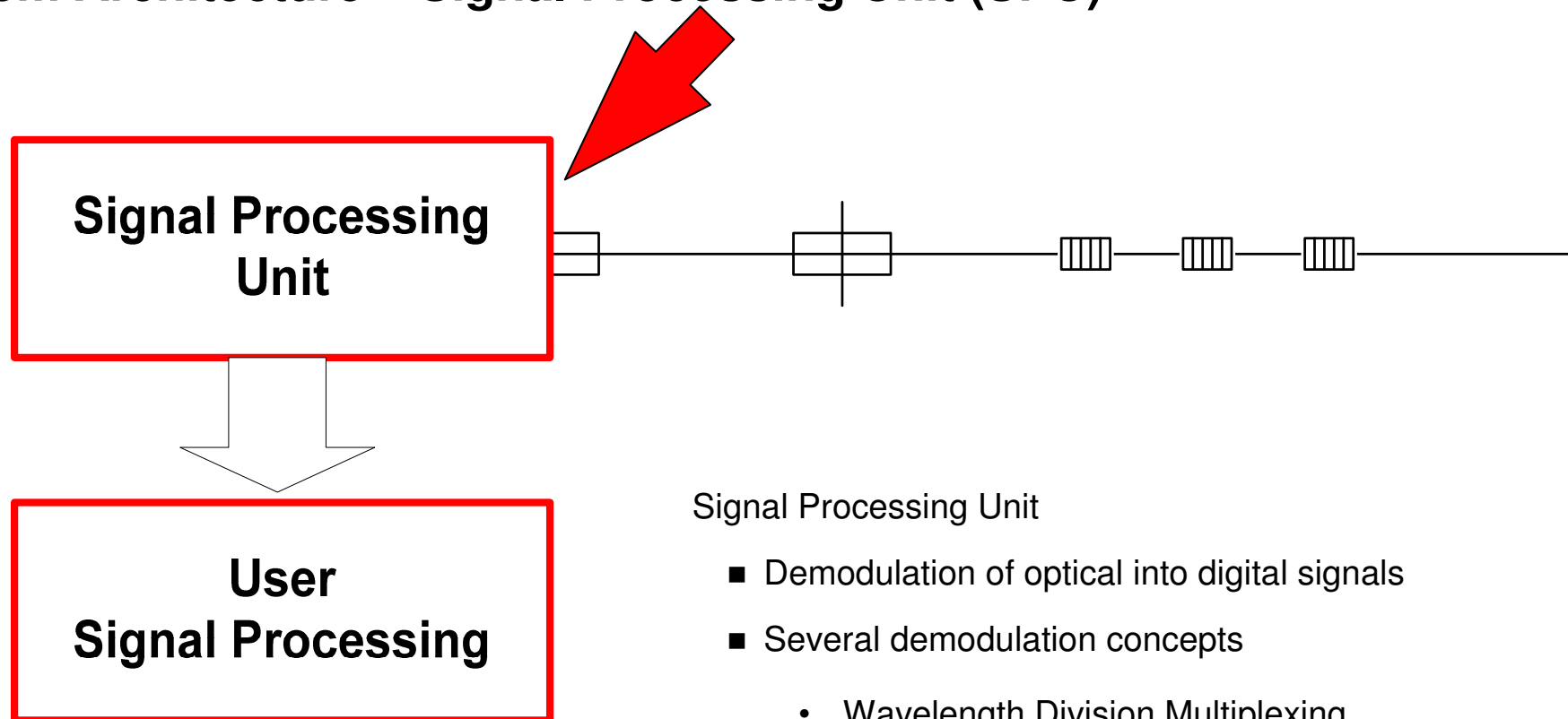


Sensor Fiber with Fiber Bragg Gratings (FBG)

- FBG's can be distributed arbitrarily along the sensor fiber
- Diameter of sensor fiber: 55- 250 μm (standard)
- Coating: Acrylate (standard), Polyimide, ORMOCER, Metallic
- Max. elongation: 1% (Standard), up to with draw tower FBG's

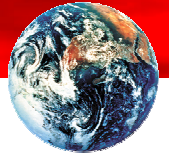


System Architecture – Signal Processing Unit (SPU)

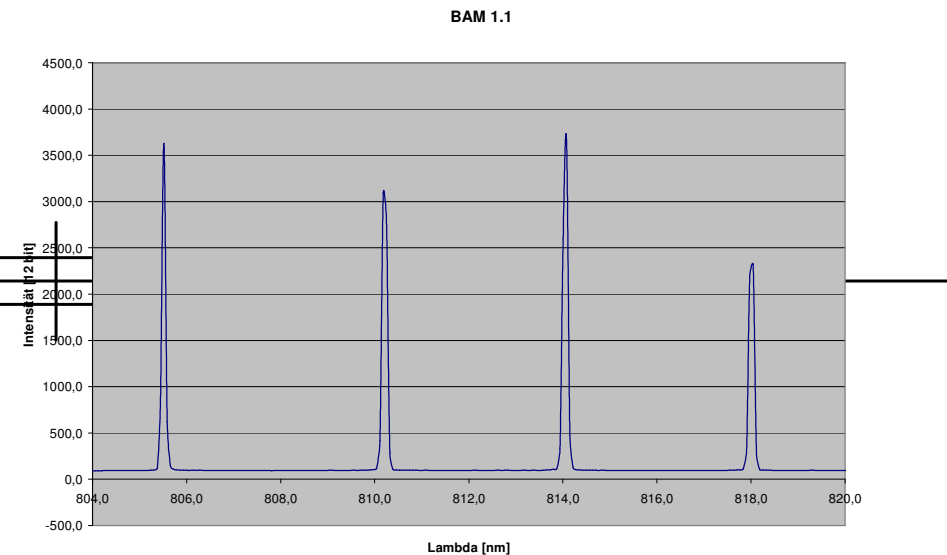
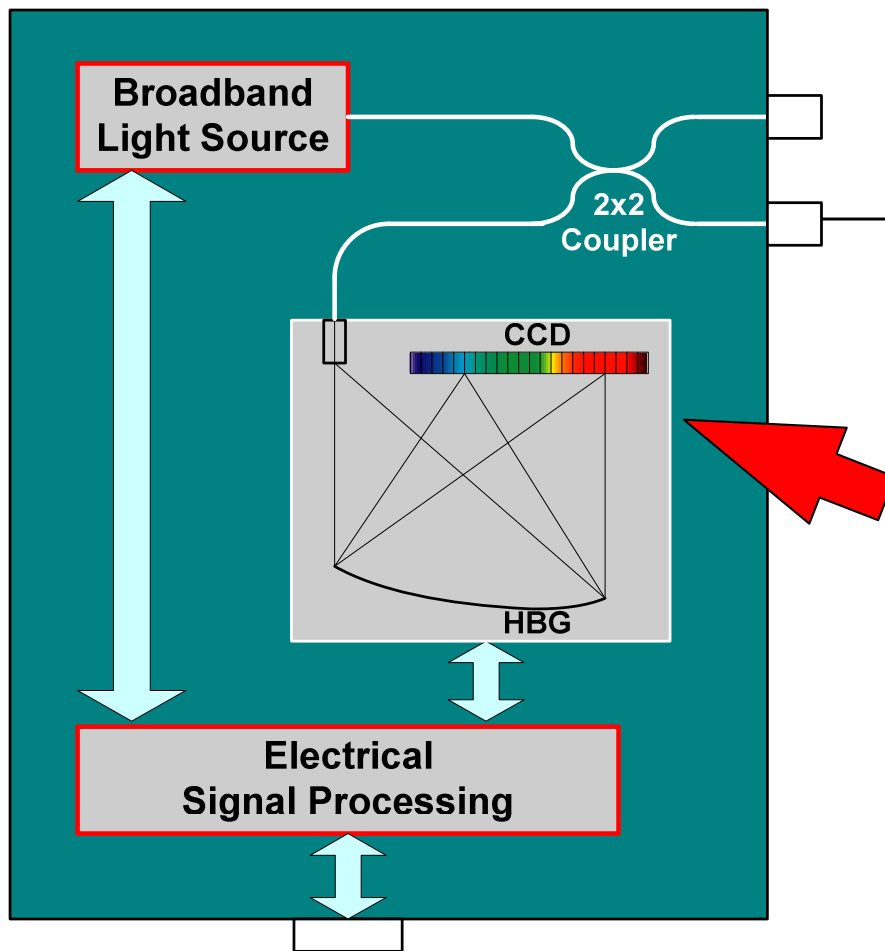


Signal Processing Unit

- Demodulation of optical into digital signals
- Several demodulation concepts
 - Wavelength Division Multiplexing
 - Time Division Multiplexing

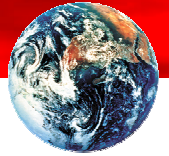


Signal Processing Unit (SPU)



Spectroscopic demodulation:

- CCD line sensor in the 800nm range
- Synchronous data sampling (photo shot)
- High sensitivity, resolution and SNR
- Mechanical stable and reliable
- Verified measurement technique



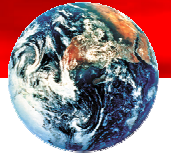
Fiber Optic Signal Processing Under Severe Environmental Conditions

800 nm Wavelength Range instead of 1500 nm

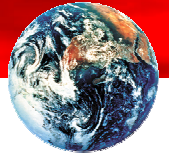
- Usage of inexpensive silicon based photodetectors
- Relaxed detector cooling
- Feasibility of small, robust and inexpensive spectrometers (industrial standard)
- Availability of double measurement range (800 nm range $DI / De = 7 \text{ pm} / 10 \text{ } \mu\text{m/m}$, in the 1500 nm range $DI/De = 15 \text{ pm} / 10 \text{ } \mu\text{m/m}$)
- Good resolution for strain and temperature monitoring (resolution $< 3\text{pm}$ or $<5 \text{ } \mu\text{m/m}$ respectively)
- High MTBF (45.000 h for X-38)

Spectrometer Design

- Synchronous data sampling (photo shot) of all FBG's (\rightarrow Very important for dynamic load monitoring)
- High speed detectors available
- Secure availability of all components in MIL standard (i.e. light source from fiberoptic gyros)



Application on X-38



Application on X-38

Objective

- Demonstration of Health Monitoring Technologies on-board of the NASA re-entry vehicle X-38

Project TETRA (Technologies for Future Space Transportation):

- Cooperation between DLR and NASA on RLV technologies
- Period: 1996 -2002
- Funded by BMBF, Bavarian State and own funds

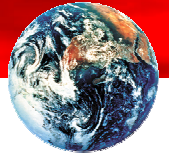
Fiber optic sensor application

- Strain and temperature measurements
- Sensing of strain and temperature measurements during ascent and decent

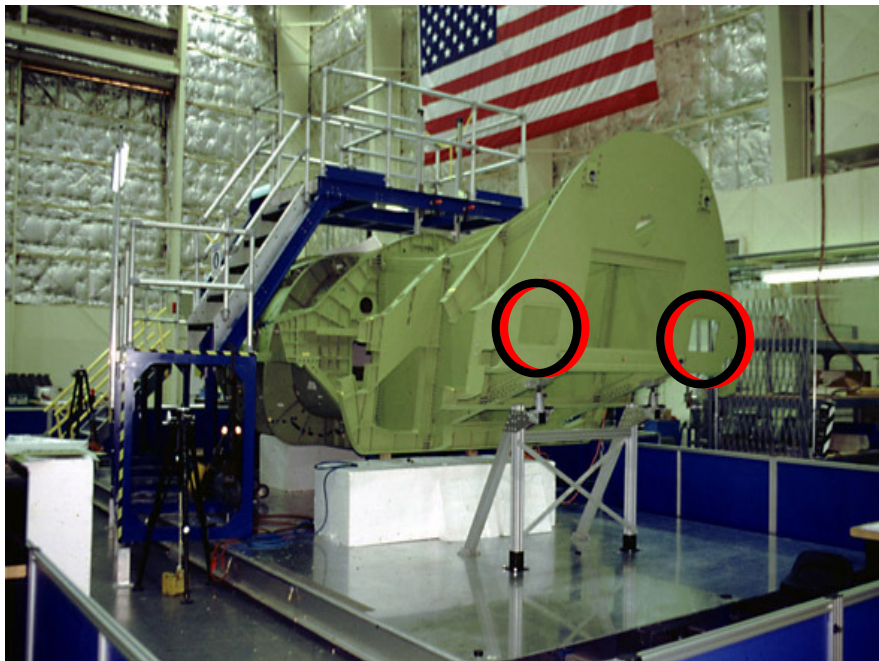


Dryden Flight Research Center ED97-43903
X-38: Crew Return Vehicle technology demonstrator re-entering Earth's atmosphere. 1997 Artist concept for NASA





Application on X-38

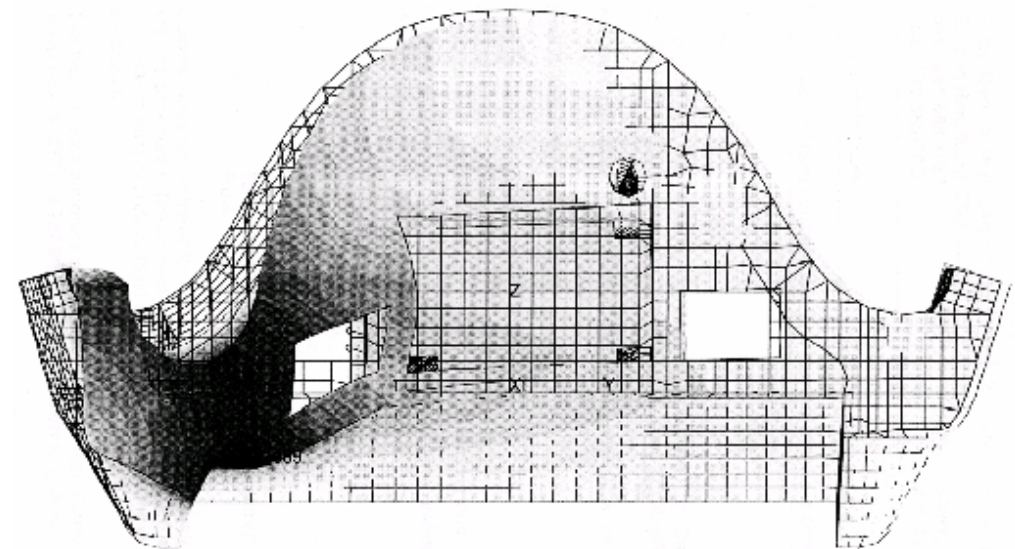


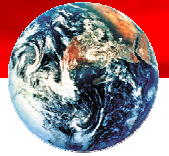
Load Measurements on the aft structure of the X-38

Strain range: -1000 $\mu\text{m}/\text{m}$ – 4000 $\mu\text{m}/\text{m}$

Temperature range -40 °C - 180 °C

Sample rate 1 Hz (static values)





X-38 Requirements for the SPU

Environmental vibration

- 20Hz 0.01g²/Hz
- 40Hz 0.160g²/Hz
- 350Hz 0.160g²/Hz
- 2000Hz 0.0005g²/Hz
- Overall 9.87grms

Thermal-vacuum

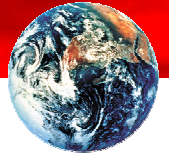
- -15°C to 50°C
- pressure from 0 bar to 1 bar

Supply Power Voltage: 28 VDC \pm 6 VDC

(Continuous and inrush resistant power supply)

EMC/EMI: SPP 30237

Data: RS 422

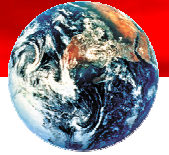


Critical Technology Developments

Signal Processing

Sensor Fiber

Verification



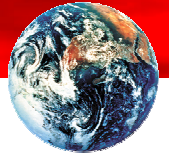
Critical Technology Developments – Signal Processing Unit

X-38 requirements

- Operation during severe environmental conditions
- Compact and rugged design for space application
- Low power consumption
- Qualified components

Compensation of environmental impact on sensing

- Vacuum
- Temperature range
- Vibration



Environmental Impact on Sensing

Vaccum

Effect

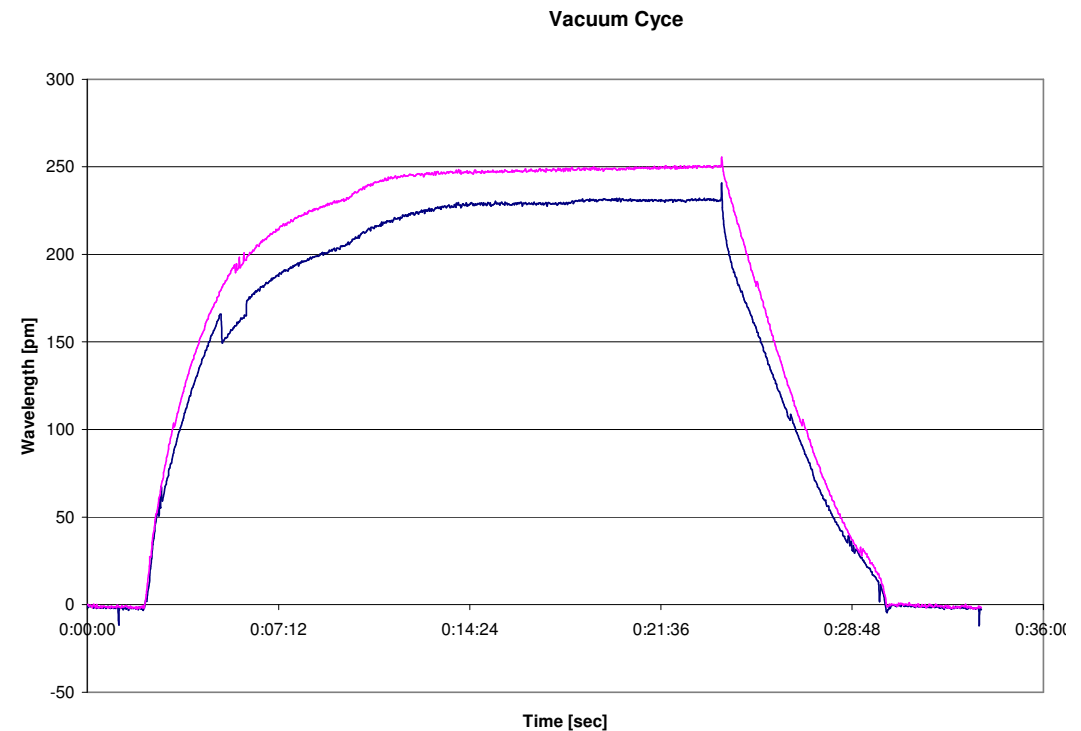
- Drift with pressure changes

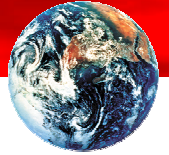
Reason

- Caused by the refractive index changes between normal pressure and vacuum.

Technical Solution

- Reference grating





Environmental Impact on Sensing

Temperature

Effect

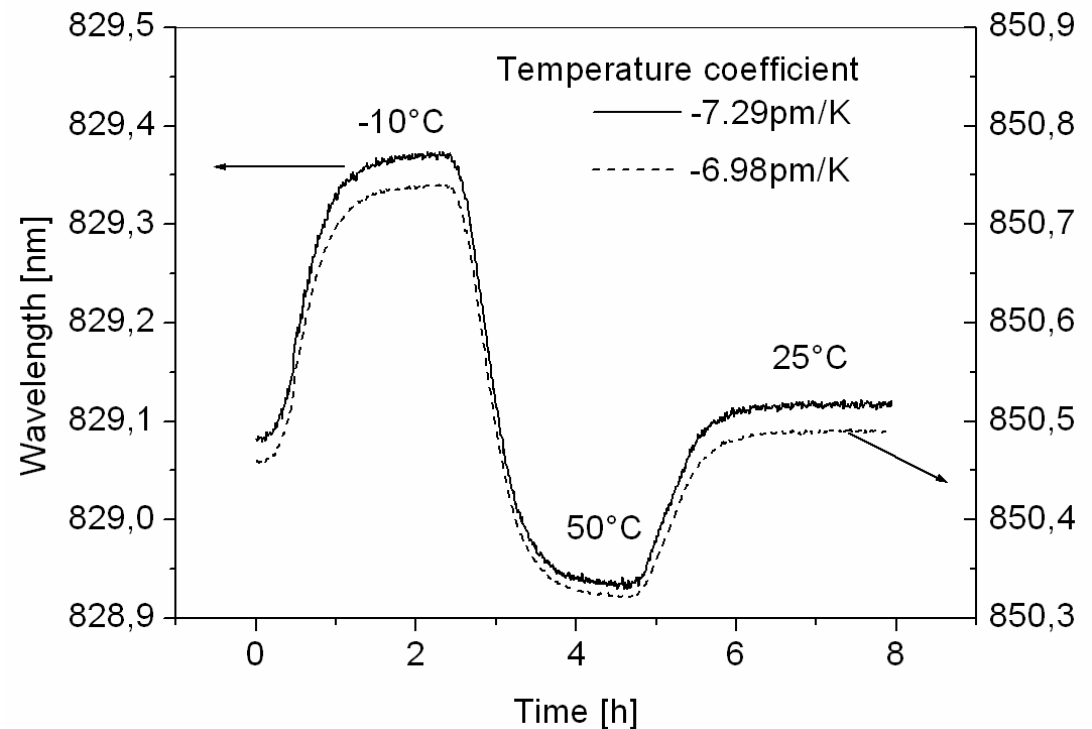
- Thermal drift of spectrometer

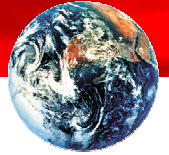
Reason

- Extension of optical path
- Thermal drift of spectrometer grating

Technical solution

- Optical bench with minimal CTE (e.g. Invar)
- Reference grating





Environmental Impact on Sensing

Vibration

Effect

- High noise due to disturbances on optical path

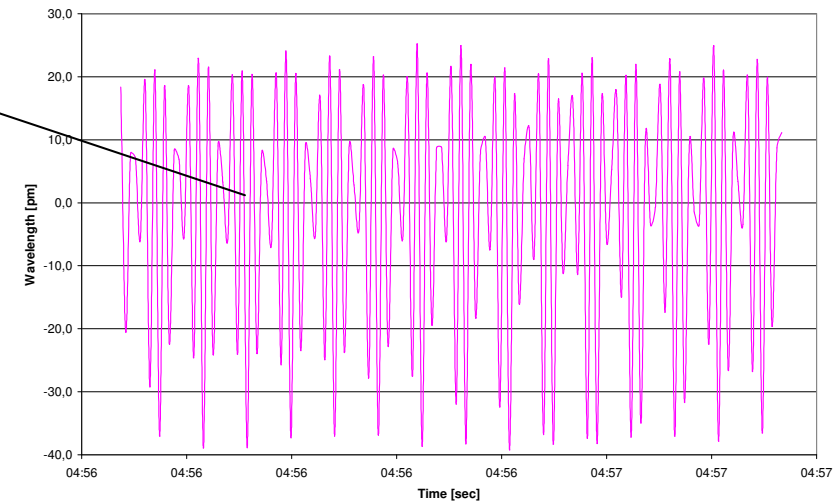
Reasons

- Vibration of optical bench
- Vibration of connectors

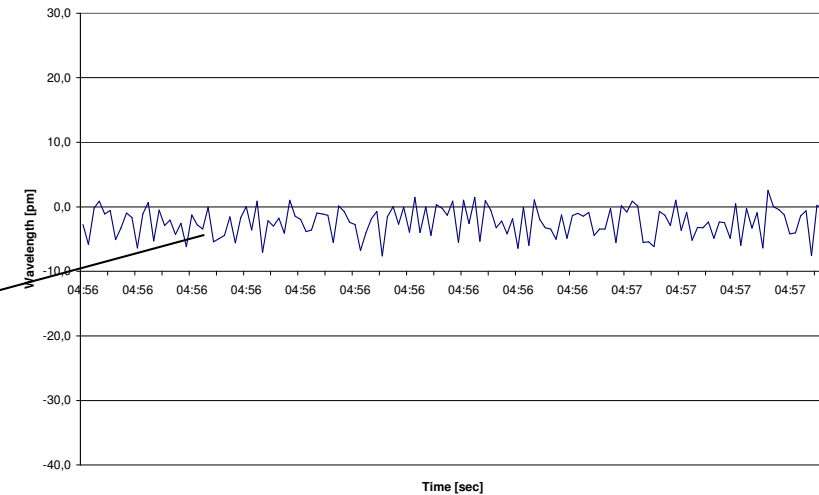
Technical solution

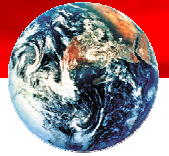
- Reference grating

w/o reference
grating



With reference
grating



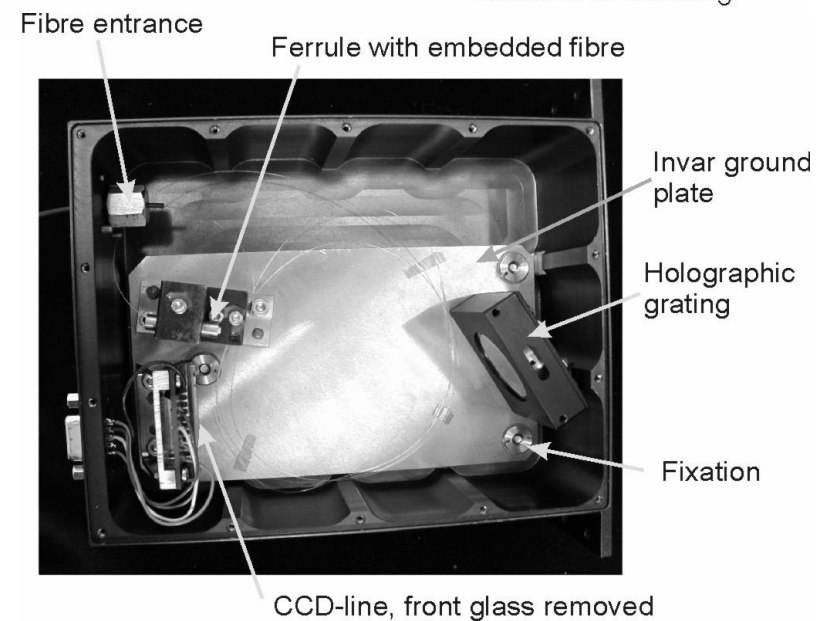
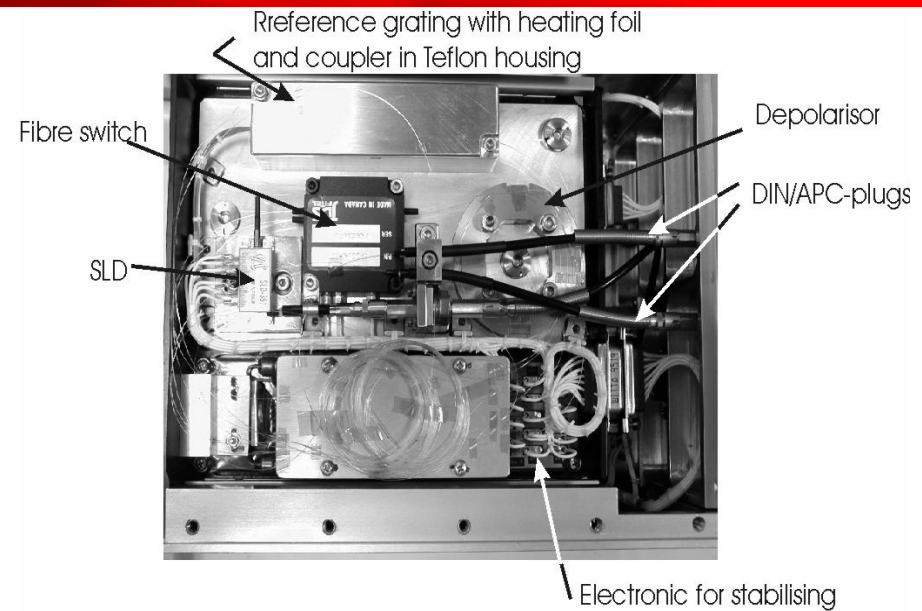


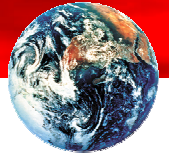
SPU Design



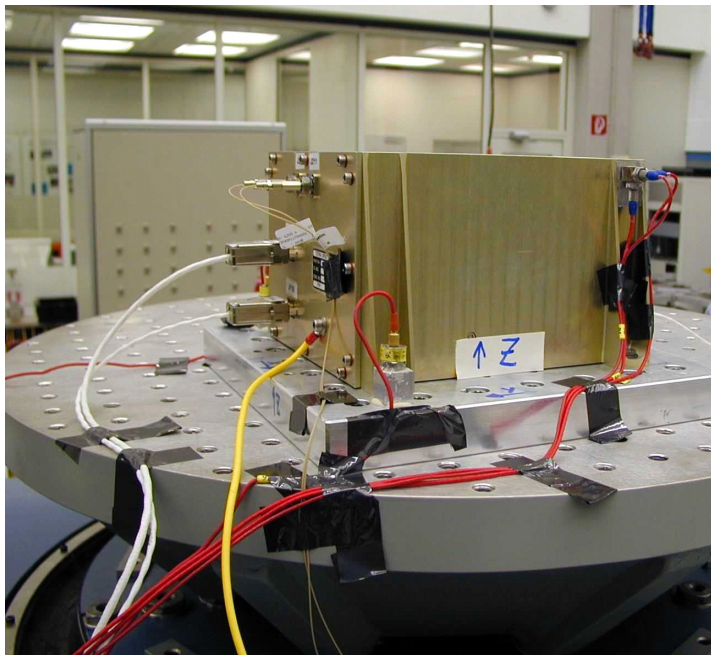
Dim: 220 x 220 x 140 mm

Mass: 5,5 kg

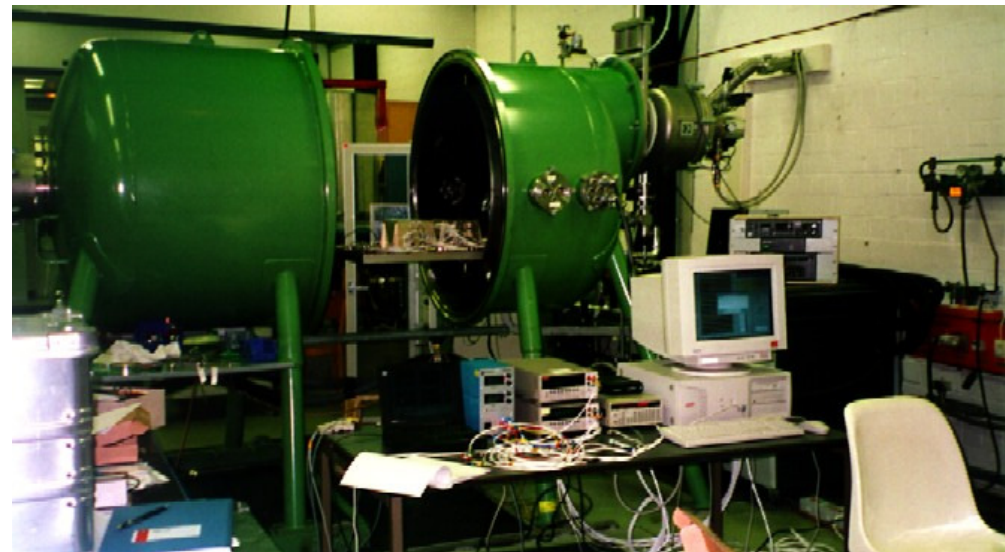




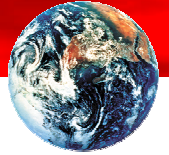
SPU Verification



Vibration test



Thermal vacuum tests



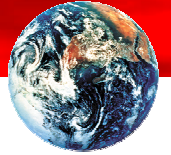
Critical Technology Developments - Sensor Fiber

X-38 requirements

- Gluing on aluminium
- Severe environment

General sensor fiber requirements

- Suitable coating for application
- Characteristica of sensor fiber
 - Quality problems with recoated sensor fibers
 - No standards for characteristic values (e.g. p_{eff} , mechanical and thermal hysteresis, etc.)



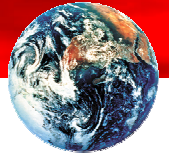
Critical Technology Development - Gluing

Problem

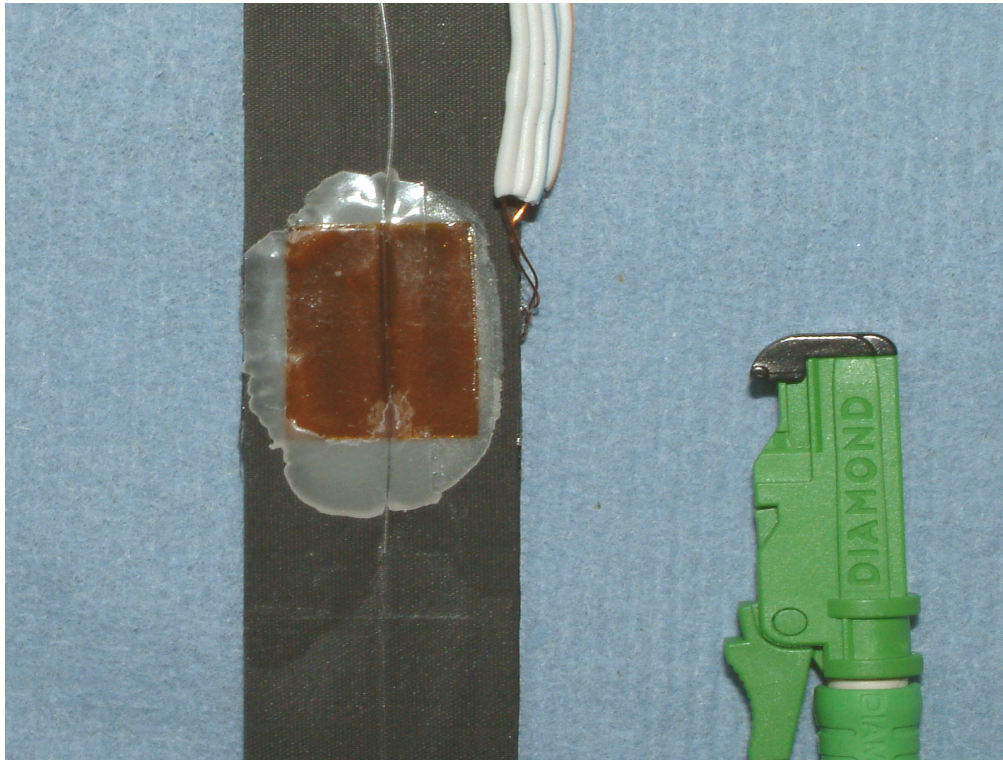
- Many criterias to be considered: coating, sample material, surface characteristics, environmental conditions
- Low mechanical transfer behaviour of coatings
- Low reproducibility

Solution

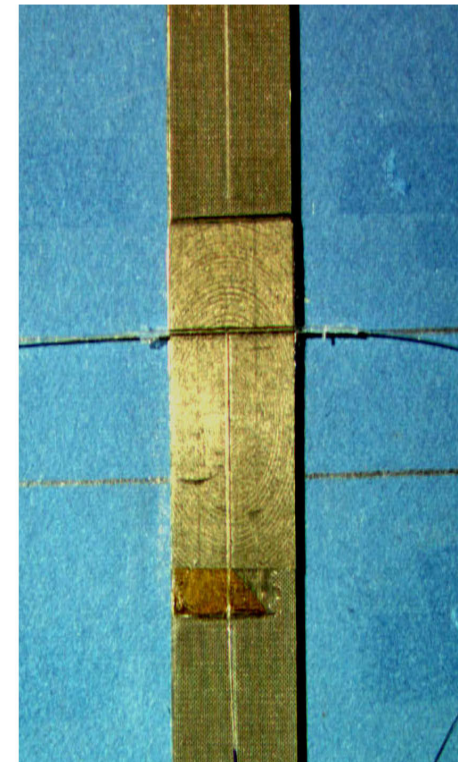
- Execution of a glue trade off
- Design of different sensor configurations, e.g. sensor pads
- Development of an application procedure inclusive tools
- Qualification tests with each FBG before gluing
- Calibration required for each sensor after gluing



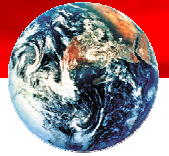
Critical Technology Development - Gluing



Fiber glued with sensor pad

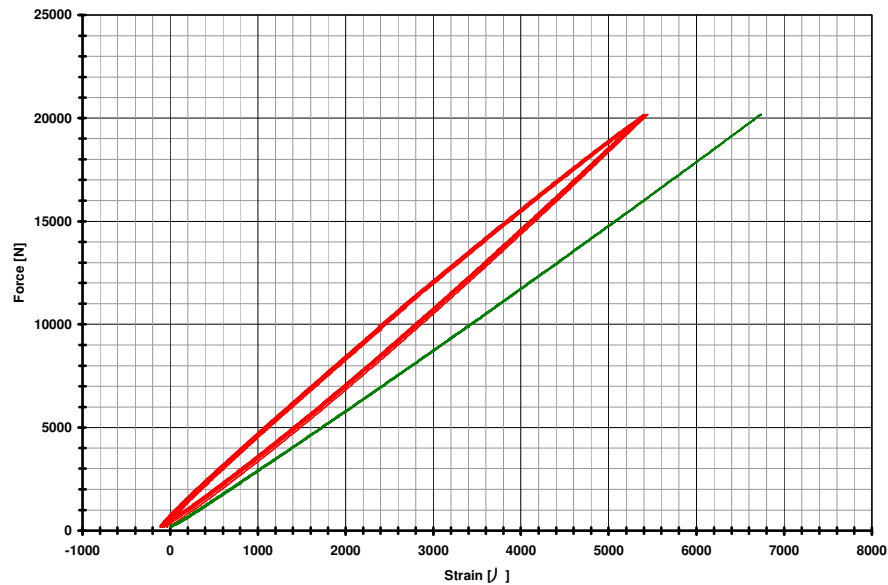


Sensor fiber glued directly

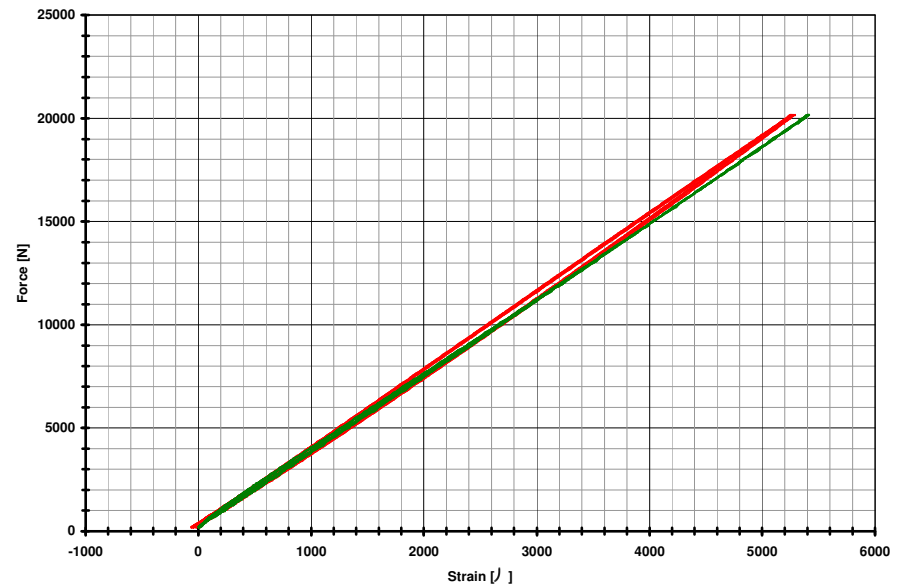


Critical Technology Development - Gluing

Bad connection



Good connection





Critical Technology Development - Coatings

Acrylate (standard)

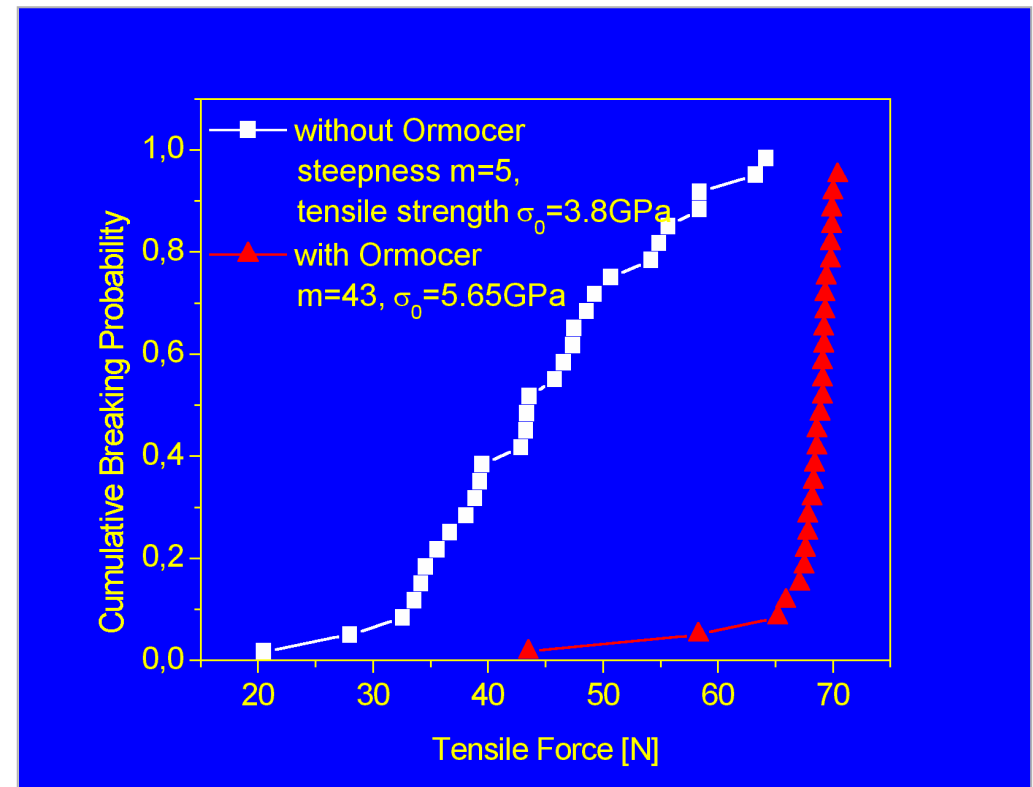
- Low mechanical and thermal stability

Polyimide

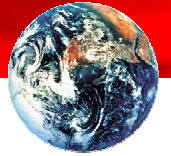
- Thermal stability $-270...+280^{\circ}\text{C}$
- Improved mechanical strength

ORMOCER® (Organic Modified Ceramics)

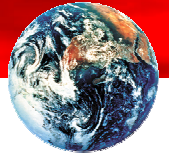
- High mechanical stability
- Thermal stability $-60...+250^{\circ}\text{C}$



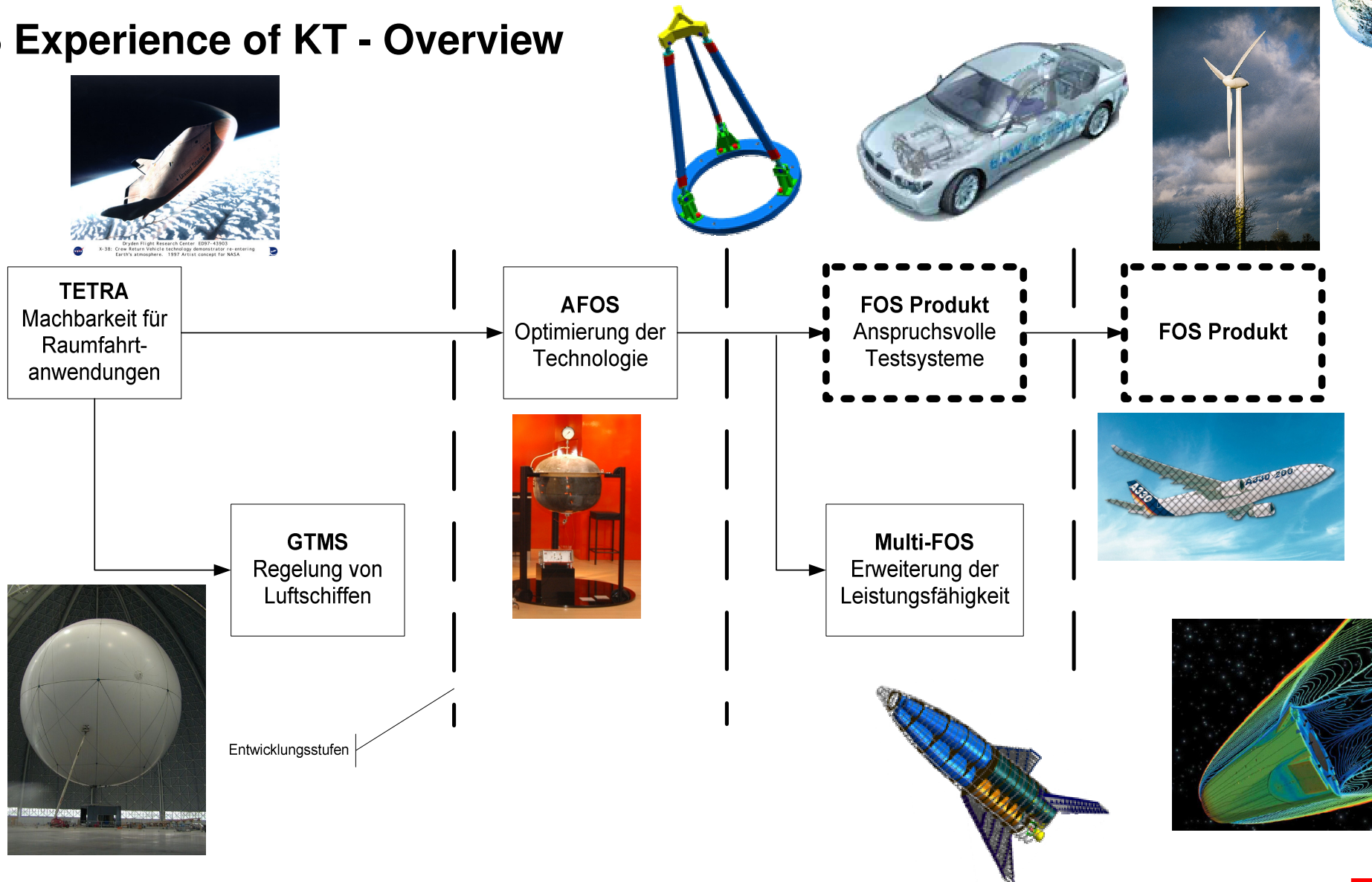
Weibull diagram: mechanical breaking probability of Polyimide coated fibres with and without ORMOCER re-coating (thickness $20\mu\text{m}$)

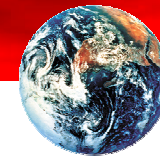


Other Applications and FOS Systems



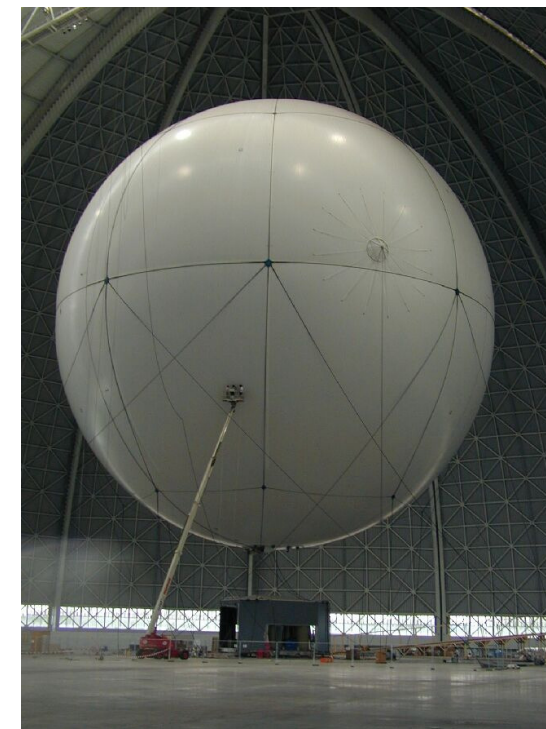
FOS Experience of KT - Overview

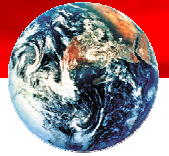




Industrial Applications

- Aviation (health monitoring)
- Airships (flight control)
- Wind turbine (process control)
- Automotive (load monitoring)

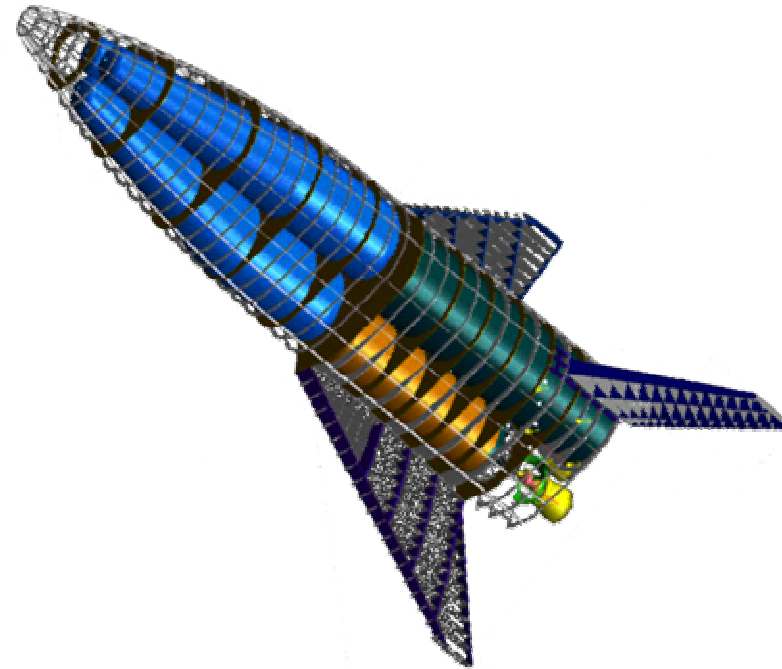
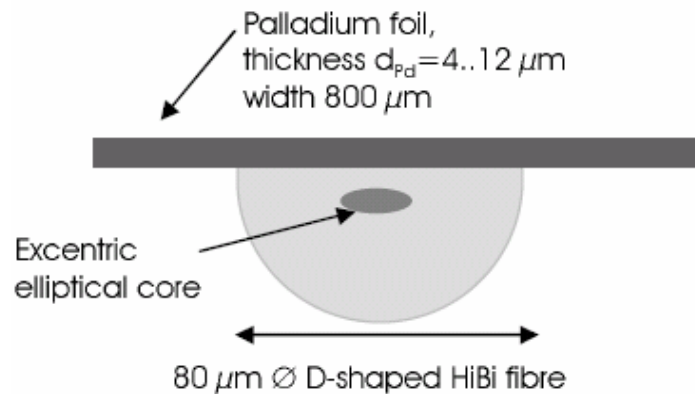




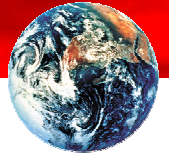
Space Application

Multipurpose Fiber Optic Sensor for RLV

- ESTEC TRP contract
- Monitoring of strain, temperature and hydrogen concentration in cryogenic hydrogen tanks
- Sensor embedded in composite

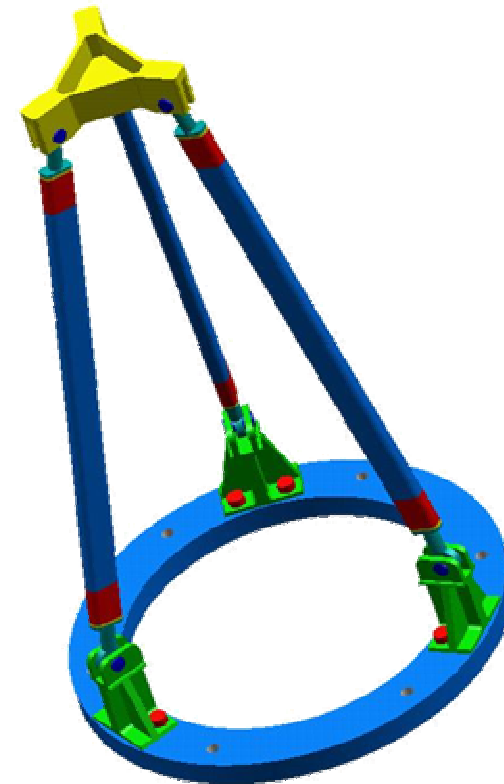


With H_2 : Pd elongation \rightarrow bending
 \rightarrow Fibre core elongation
 \rightarrow Bragg wavelength shift

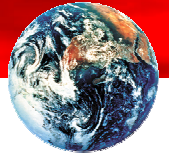


Space Application

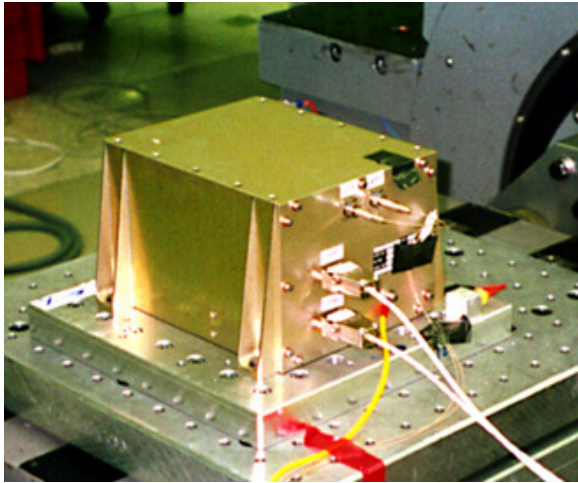
Structural Deformation Control of high precision optical bench (ESTEC contract)



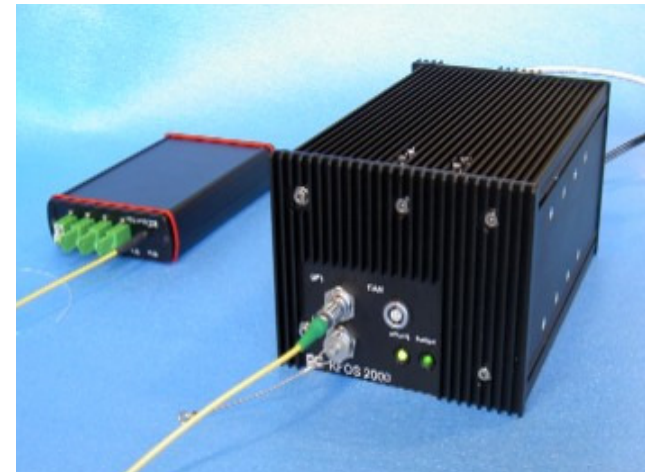
Cassegrain tripod



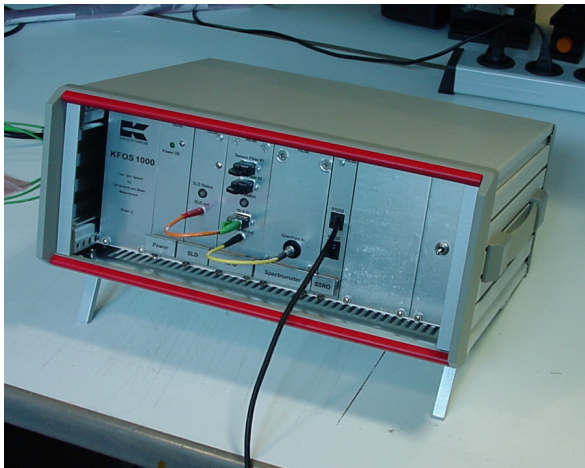
Signal Processing Units For Different Applications



Space



High-end test system, e.g. for
Automotive



Laboratory / industry



Conclusion

- Robust and compact signal processing unit suitable for almost all environments
- High mechanical strength and lifetime of sensor fibers
- Compensation of environmental impact on signal
- High Reliability and availability due to design and selection of components (high MTBF, remote monitoring, Built-In-Test)
- Easy to adapt to application requirements (interfaces, data format, signal processing)